| $\begin{aligned} & \text { Q=QUESTIO } \\ & \text { A=ANSWER } \end{aligned}$ | question_description answer_description | question_explanation answer_explanation | question_type answer_isright | question_difficulty answer_position |
| :---: | :---: | :---: | :---: | :---: |
| Q | The average value of the angular position of the satellite with reference to perigee is given by |  | M | 1 |
| A | True Anamoly |  | 0 | 1 |
| A | Mean Anamoly |  | 1 | 2 |
| A | Inclination |  | 0 | 3 |
| A | Semi-major axis |  | 0 | 4 |
| Q | The point where orbit crosses equitorial plane going from North to South is |  | M | 1 |
| A | Descending Node |  | 1 | 1 |
| A | Ascending Node |  | 0 | 2 |
| A | Apogee |  | 0 | 3 |
| A | Perigee |  | 0 | 4 |
| Q | Semi-Major axis of an orbit defines |  | M | 1 |
| A | Size of the orbit |  | 1 | 1 |
| A | Shape of the orbit |  | 0 | 2 |
| A | Inclination of the orbit |  | 0 | 3 |
| A | Location of the satellite in the orbit If R is radius of the earth then apogee |  | 0 | 4 |
| Q | height is given by |  | M | 1 |
| A | ha=R-ra where ra= a $11+\mathrm{e}$ ), |  | 0 | 1 |
| A | ha=R-rp where $\mathrm{rp}=\mathrm{a}(1-\mathrm{e})$, |  | 0 | 2 |
| A | $\mathrm{ha}=\mathrm{ra}-\mathrm{R}$ where $\mathrm{ra}=\mathrm{a}(1+\mathrm{e})$, |  | 1 | 3 |
| A | ha $=r p-R$ where $r p=a(1-e)$, |  | 0 | 4 |
| Q | If $R$ is radius of the earth then perigee height is given by |  | M | 1 |
| A | $\mathrm{hp}=\mathrm{R}-\mathrm{ra}$ where $\mathrm{ra}=\mathrm{a}(1+\mathrm{e})$, |  | 0 | 1 |
| A | $\mathrm{hp}=\mathrm{R}-\mathrm{rp}$ where $\mathrm{rp}=\mathrm{a}(1-\mathrm{e})$, |  | 0 | 2 |
| A | $\mathrm{hp}=\mathrm{ra-R}$ where $\mathrm{ra}=\mathrm{a}(1+\mathrm{e})$, |  | 0 | 3 |
| A | $\mathrm{hp}=\mathrm{rp}-\mathrm{R}$ where $\mathrm{rp}=\mathrm{a}(1-\mathrm{e})$, |  | 1 | 4 |
| Q | In C band the down link frequency is |  | M | 1 |
| A | 6 GHz |  | 0 | 1 |
| A | 4 GHz |  | 1 | 2 |
| A | 11 GHz |  | 0 | 3 |
| A | 14 GHz |  | 0 | 4 |
| Q | For satellite communicationwhich band of the following is not used? |  | M | 1 |
| A | Ku |  | 0 | 1 |
| A | MF |  | 1 | 2 |
| A | Ka |  | 0 | 3 |
| A | C |  | 0 | 4 |
| Q | The time period in which a particular satellite must be launched is called as |  | M | 1 |
| A | Orbital time period |  | 0 | 1 |
| A | Launch Window |  | 1 | 2 |
| A | Lapsed time |  | 0 | 3 |
| A | Mean time |  | 0 | 4 |


| Q | The east and west limits on the geostationary arc visible from any given earth station is called as | M |  | 1 |
| :---: | :---: | :---: | :---: | :---: |
| A | Look angles |  | 0 | 1 |
| A | Limits of Visibility |  | 1 | 2 |
| A | Nadir angle |  | 0 | 3 |
| A | Range of satellite |  | 0 | 4 |
| Q | Sun Synchronous orbits are normally | M |  | 1 |
| A | Polar Orbits |  | 1 | 1 |
| A | Geostationary orbits |  | 0 | 2 |
| A | Parking Orbits |  | 0 | 3 |
| A | Transfer orbits |  | 0 | 4 |
| Q | Line joining center of the sun center of the earth and first point of aries at spring equnox is called as | M |  | 1 |
| A | Line of nodes |  | 0 | 1 |
| A | Line of apsides |  | 0 | 2 |
| A | Line of aries |  | 1 | 3 |
| A | boresight |  | 0 | 4 |
| Q | The spy satellites are normally in the | M |  | 1 |
| A | LEO |  | 1 | 1 |
| A | MEO |  | 0 | 2 |
| A | HEO |  | 0 | 3 |
| A | Geostationary orbits |  | 0 | 4 |

